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JPRS: 30,678

18 June 1965

R. Wilson

TT: 65-31290

STUDIES IN HEAT RESISTANCE OF CAST ALUMINUM ALLOYS

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This report examines the heat, resistance of alloys of the septem Al- Cu-Mg-Si-Mn-Cr-Ti and presents the Jundamental concepts used as a hasis for the development of the AL 20 alloys -> 143

CHAPTER VII

EXAMINATION OF THE HEAT RESISTANCE OF ALLOYS OF THE SYSTEM
Al-Cu-Mg-Si-Mm-Cr-Ti AND OTHERS

1. Fundamental Concepts Used As Basis For the Development of the AL20 Alloy*

It was shown in the study of V. A. Livanov, D. A. Petrov, V. P. Kozlovskaya and Ye. I. Kutaytseva, carried out in 1944, that an alloy of the system Al-Cu-Mg on a base of solid aluminum solution and containing copper and magnesium in quantities corresponding to the ratio of these elements in the S(Al2CuMg) phase has the highest short time heat resistance of all the alloys of this system at temperatures of 250-300°C. Therefore an alloy with a content of up to 5% copper and 1.7% Mg was taken as the basis for the development of a new super alloy intended for prolonged operation at 250-300°C. A higher content of these components will lead to a reduction of the alloy!s ductility.

Alloys of this system cannot have good casting properties because their base is a solid solution. A definite amount of a eutectic (not less than 30%) is needed for improving their casting properties. The examined alloys were alloyed with silicon for this purpose. It was assumed that elementary silicon would sharply lower the alloy's heat resistance, therefore it should participate in the formation of stable phases of a complex composition.

The investigations showed that the most heat resistant phases containing a silicon are the phases Al_5SiFe , Al_4Si_2Fe and π ($Al_8Si_6Mg_3Fe$). In order to form these phases, iron in ratios corresponding to the composition of the Al_5SiFe phase or the phases indicated above has to be jointly introduced with the silicon into alloys of the system Al-Cu-Mg. The content of each of them should not exceed 2% because in a contrary case the alloy's ductility is lowered.

^{*} In addition to I. F. Kolobnev, D. A. Petrov, G. V. Zakharova, A. V. Korobkov and V. N. Ozepetskovskiy participated in the development of the AL20 alloy.

Alloying with small additions of manganese, chromium and titanium was made use of for reducing the structure and increasing the alloy's ductility. These addition can also cause an increase in the heat resistance.

The chemical composition of the examined alloys is given in Table 70. The test specimens of standard shape were prepared by the usual method with the use of A00 grade aluminum. The test results are given in Table 71. The ultimate strength of the alloy increases in both the cast as well as the heat treated states with an increase in the copper and magnesium content. But the test samples of alloys of the system Al-Cu-Mg are more stable in the heat treated condition than in the cast. It should be noted that alloys containing silicon and iron in a ratio corresponding to the composition of the Al₅SiFe phase have a high heat resistance in the cast state. Alloys with a higher silicon content have a decreased heat resistance which is associated with the presence of elementary silicon in them.

Table 70
Chemical composition of the examined alloys

(1)		(2)	.Содерж	ание. %. о	стальное в.	люминий	· · · · · · · · · · · · · · · · · · ·	
Номер сплава	Cu	Mg	SI	Fe	NI	Ti	Cr_	Mn
1 2 3 4 5 6 7 8 9 10 11 12 13 14	2,22 3,11,4,46 2,11 3,32 4,6 3,78 4,44 3,53 4,20 3,03 3,03 3,03 3,22 3,18	0,76 1,24 1,67 0,90 1,32 1,75 1,25 1,65 1,30 1,66 1,27 1,24 1,14 1,21	 1,35 1,81 1,42 1,25 1,26 1,27 1,37 2,85 5,37 10,39 10,25		 		— — — — 0,2 0,25 — цированнь	

(5) Примечания. 1. В сплавы № 1—9 медь и магний добавляли в соотношениях образования фазы S (Al_{*}MgCu).
2. В сплавы № 10—14 железо и кремний добавляются в соотношениях образования фаз Al_{*}SiFe.

Key: 1-Alloy number; 2-content, % with aluminum remainder; 3-modified; 4-unmodified; 5-Notes. 1. Copper and magnesium was added to alloys 1-9 in ratios corresponding to the formation of the S(Al2MgCu).phase. 2. Iron and silicon was added to alloys 10-14 in ratios corresponding to the formation of the Al2SiFe phase.

The data in Table 72 attest to the fact that alloys 7, 8, 9 and 10 have completely satisfactory mechanical properties at room and elevated temperatures in the cast and heat treated states; alloy No 7 is the most ductile, it was therefore taken as the base for the new alloy under the designation AL20 (V14A) and it was checked for ultimate strength at 250°C.

The following composition for the AI20 alloy was established as the result of an all inclusive investigation: 3.5-4.5% Cu, 0.7-1.2% Mg, 1.5-2% Si, 1.2-1.7% Fe, 0.2-0.3% Mn, 0.1-0.18 Cr and 0.1-1.18% Ti with the remainder being aluminum.

In ratio of the basic components of copper, magnesium and silicon the AL20 alloy is found [2] in the region $\lambda + s - \text{Mg}_2\text{Si} - \text{CuAl}_2$ of the two dimensional cross section diagram of the tetrahedron Al-Cu-Si-Mg (at 90% Al) shown on Figure 91.

It was established that the AL20 alloy with an addition of manganese, chromium and titanium has a higher heat resistance but the ductility of this alloy is lower than that with an addition of titanium. Changes in small additions do not have any essential effects upon the engineering properties of the alloy.

Table 71

Mechanical properties of the alloys as a function of temperature and composition

			(2)	Coc	тоянне	сплава			
(1)		(3) инто	(5)		іное ії с состарен	стественио ное		гос 11 но состарен	кусственно
	механи 4) свой при 2	ства	длитель- ность испытаний при 200° С	CBOD	механические 7) свойства при 20° С		механия 10 желения 10 желения	ства	длитель-(1 ность (1 испытаний при 300° С
Номер сплава	а _р кг/жж ³	8. %	и напряже- нин 6 кг/мм ^е час.	KS/WW ₈	δ. %	при 300° С и напряже- ини 6 кг/мм ⁹ час.	о _ь кг/мм²	გ. %	и напряже- нии 6 кг/мм ^в час.
1 2 3 4 5 6 7 8 9 10 11 12 13 14	12 13,7 15,4 18,8 19,9 20,0 18,8 20,4 22,2 19,9 16,9 18,7 16,7 18,9	1,37 0,75 0,27 2,1 1,9 0,9 1,57 1,33 1,0 0,45 0,45 0,5	1,83 14,3 23,3 27,1 66,7 94,5 131 148 128 157 34 38 28 59	19,6 20,8 22,5 21,4 21,6 22,7 25,9 23,4 25,6 22,7 25,6 19,85 20,6	5,58 1,3 1,43 3,5 2,4 0,96 1,8 2,18 1,4 0,8 1,1	6 33,5 45 31,5 70 82,7 113,3 138 115 134 33,3 31,8 15	19,8 23,6 22,5 25,1 24,6 28,4 28 34,4 26 27,4 20,9 24,0 17,7 17,0	2,4 1,2 0,75 	9 44,6 72,6 65,7 69,3 78,7 103 121 108 116 20,8 20 27,3 16

Key: 1--Alloy number; 2--Condition of alloy; 3--Cast; 4--mechanical properties at 20°C; 5--duration of testing at 200°C and load of

Key to Table 71 (continued)

6 kg/mm², hours, 6-hardened and naturally aged; 7-mechanical properties at 20°C; 8-duration of testing at 300°C and load of 6 kg/mm², hours; 9-hardened and artificially aged; 10-mechanical properties at 20°C; 11-duration of testing at 300°C and load of 6 kg/mm², hours.

The AL20 alloy is almost a unique alloy which contains up to 1.7% Fe as the basic alloying constituent. Hence its application in industry promotes a wider use of secondary alloys and industrial scrap which has been contaminated with iron.

Tables 72-74 give the mechanical characteristics of the AL20 alloy as a function of its condition. The heat resistance of the AL20 alloy is adequately high in both the cast as well as in the hardened and artificially aged states. It should be noted that during short time tensile testing the alloy in heat treated condition has a superiority over the cast alloy only up to 250°C. At higher temperatures the tensile strength of the AL20 alloy in both states is practically identical whereas the yield stress in the heat treated condition is higher at all test temperatures. This is a characteristic peculiarity for all multicomponent alloys with a strongly alloyed solid solution.

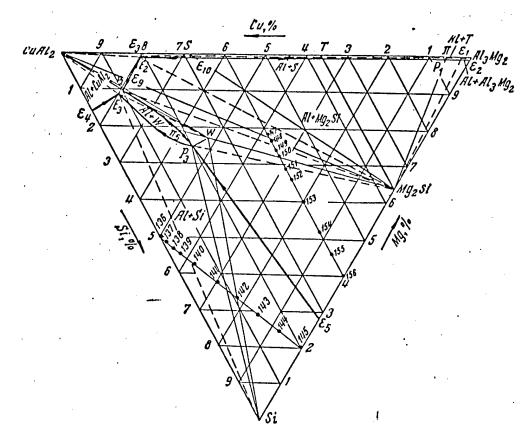


Figure 91. Two dimensional diagram for 90% Al of the tetrahedron Al-Si-Cu-Mg.

Duration of testing in hours of individually cast test specimens of AL20 alloy

		(9) Состояни			
(1)	(2)	литое	(5)	: закаленное 1	и искусствен	ю состаренное
Номер сплава	(3) механически при 2	іе свойства 0° С	длительность испытаний при 250° С	механически при 2	(7) длительность испытаний при 250° С	
	те/мм ³	8. %	о _в = 9 кг/мм³ час.	а _р кг/жм³	8. %	о _b = 8 кг/мм ^в час.
1 2 3 4 5 6 7 8	- 20 18,2 18,3 — — —	1,8 1,3 1,6 —	100* 100* 100* 96 100*	27,5 28,4 28,4 — — — —	1,3 0,7 1,3 — —	100* 100* 100* 100* 100*

^{(8) •} Образцы были сняты без разрушения и переведены на следующее напряжение.

Key: 1--Alloy number; 2--Cast; 3--mechanical properties at 20°C; 4--duration of testing at 250°C, $\sigma_b = 9 \text{ kg/mm}^2$, hours; 5--hardened and artificially aged; 6--mechanical properties at 20°C; 7--duration of testing at 250°C, $\sigma_b + 8 \text{ kg/mm}^2$, hours; 8--* The test specimens were removed without failure and then carried over to the following load; 9--Condition of alloy.

Table 73

Short time testing of AL20 alloy

1)					(2)	Cod	тояни	е спла						
8. O		(3	3)	литоя	l .		<u> </u>	зака.	тенный	н нс	кусст	венно с	остарен	ный •••
Температура испытания. "С	о _в кг/иж ⁸	S _K K2/MM ⁸	. % %	\$· ÷	onp K2/KM	⁹ 0,2 кг/им ⁸	Е, кг/жж	^б ь кг/жж³	S _K Ke/MM ³	8. %	رد. %	σ _{np} κε/אκ*	³ 0.2 кс/жж	Е, кг/мм
20 00 50 00 50	16,2 15,0			0,3 0,1 0,1 0,2 0,6 2,1	5,0 7,5 7,0 6,8 4,2 2,9	14,40 12,20	7635 6855	21,75 20,6 20,8 14,9	20,8 21,2	0,9 0,82 1,5 2,6	0,9 1,1 0,8 1,4 2,75 7,1	10,7 10,45 10,2 10,0 6,7 6,5	16,6 17,9 13,0	7070 7035 6670 6590 6300 6020

Key: 1-Test temperature; °C; 2-condition of alloy; 3-cast; 4-hardened and artificially aged.

Ultimate strength of AL20 and other alloys at 300°C (Individually cast test specimens)

(1)	(2) Длительность испытаний до разрушения при напряжении 6 кг/мм³, час., в состоящия							
Марка сплава	(3)	авкаленном	закаленном и состаренном					
АЛІ	. 89	125*	114					
АЛЗ	5	4	4					
АЛ5	5	8	11					
АЛ4	1	(6) Разрушился	при нагружении					
АЛ20	140	125*	115					

(7) Примечание. Сплав АЛ20 термически обрабатывали по режиму: нагрев под закалку при 515° С в течение 3 час.; закалка в воде; старение при 250° С в течение 10 час.

Key: 1-Designation of alloy; 2-Duration of testing to failure at a load of 6 kg/mm², hours, in condition; 3-cast; 4-hardened; 5-hardened and aged; 6-Failed at load of; 7-Note. The AL20 alloy was heat treated according to the regime: heating to below hardening at 515°C for 3 hours; water quenching; aging at 250°C for 10 hours; 8-* Removed without failure.

2. Heat Treatment Conditions For the AL20 Alloy

The optimum heat treatment conditions were selected not for the purpose of obtaining casting with a maximum strength but rather for assuring minimum volumetric changes in the parts during their service at elevated temperatures. In this case hardening and aging with the attainment of a maximum strength are harmful.

The mechanical properties as a function of heat treatment conditions are given in Table 75. The best heat treating regime is shown, in Figure 92.

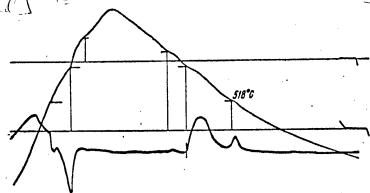


Figure 92. Heating and cooling curve for the AL20 alloy.

^{(8) •} Сняты без разрушения.

The heating to below hardening was gradual: 2-3 hours at 500°C, then 2-3 hours at 515°C. Under conditions of slow heating to 525°C one step heating will also assure high properties. Water quenching was at 80-100°C. Aging at 250°C for 10 hours.

This heat treating regime assures (for individually cast test specimens) a tensile strength of 25 kg/mm² and a relative elongation of 1.5%.

Table 75

Mechanical properties of AI20 alloy as function of heat treatment conditions

(1)	(2)Механическ	не свойства .
Режим термической обработки	σ _b , кг/мм³	გ. %
(3) Литой	18,3	1,1
(4) Литой + отжиг при 250° С в течение 3 час	18,2	1,0
(5) Литой + отжиг при 250° С в течение 35 час.	21,4	0,7
(6) Нагрев под закалку при 520° С в течение 3 час	29,5	1,3
(7) Нагрев под закалку при 520° С в течение 10 час	29,3	1,3
(8) Нагрев под закалку при 520° С в течение 10 час. + старение при 250° С в течение 10 час. с.	25.7	1,4
(9) Нагрев под закалку при 520° С в течение 10 час. + старение при 250° С в течение 10 час	27,5	0,9
(10) Нагрев под закалку при 515° С в течение 10 час	29,5	1,13
(11) Нагрев под закалку при 515° С в течение 10 час. + старение при 250° С в течение 5 час	27,2	1,3
(12)Нагрев под закалку при 515° С в течение 10 час. + старение при 250° С в течение 10 час.	24,1	1,0
(13) Нагрев под закалку при 515° С в течение 10 час. + старение при 300° С в течение 10 час	19,2	1,6
(14) Нагрев под закалку при 515° С в течение 10 час. + старение 170° С в течение 10 час	27,7	0,6
(15) Нагрев под закалку при 515° С в течение 10 час. + старение 170° С в течение 15 час	33,3	0,8
(16) Нагрев под закалку при 515° С в течение 10 час. + старение при 150° С в течение 10 час	23,8	1,5
(17) Нагрев под закалку при 515° С в течение 10 час. + старение при 150° С в течение 15 час	29,2	1,7
(18) Нагрев под закалку при 515° С в течение 13 час. + старение при 150° С в течение 50 час	33,5	0,4
(19) Нагрев под закалку при 515° С в течение 3 час. + старение при 250° С в течение 5 час	25,8	1,1
(20) Нагрев под закалку при 515° С в течение 3 час. + старение при 250° С в течение 10 час.	23,5	1,2
(21) Ступенчатый режим закалки: нагрев при 500° С в течение 3 час. В 515° С 3 час. + старение при 250° С в течение 5 час	31,2	1,1
(22) Ступенчатый режим закалки: нагрев при 500° С в течение 3 час. + 520° С в течение 3 час. + старение при 150° С в течение 35 час	36,3	1,5

Key: (Table 75)

1 -- Heat treatment conditions; 2 -- Mechanical properties; 3 -- Cast; 4-Cast + annealing at 250°C for 3 hours; 5-Cast + annealing at 250°C for 35 hours; 6-Heating to below hardening at 520°C for 3 hours; 7—Heating to below hardening; at 520°C for 10 hours; 8—Heating to below hardening at 520°C for 10 hours + aging at 250°C for 10 hours; 9-Heating to below hardening at 520°C for 10 hours + aging at 250°C for 10 hours; 10-Heating to below hardening at 515°C for 10 hours; ll-Heating to below hardening at 515°C for 10 hours + aging at 250°C for 5 hours; l2-Heating to below hardening at 515°C for 10 hours + aging at 250°C for 10 hours; l3-Heating to below hardening at 515°C for 10 hours + aging at 250°C for 10 hours 300°C for 10 hours; 14-Heating to below hardening at 515°C for 10 hours + aging at 170°C for 10 hours; 15-Heating to below.hard-ening at 515°C for 10 hours + aging at 170°C for 15 hours; 16-heating to below hardening at 515°C for 10 hours + aging at 150°C for 10 hours; 17-heating to below hardening at 515°C for 10 hours + aging at 150°C for 15 hours; 18-Heating to below hardening at 515°C for 13 hours + aging at 150°C for 50 hours; 19-Heating to below hardening at 515°C for 3 hours + aging at 250°C for 5 hours; 20-Heating to below hardening at 515°C for 3 hours + aging at 250°C for 10 hours; 21--Step by step hardening regime; heating at 500°C for 3 hours. At 515°C for 3 hours + aging at 250°C for 5 hours; 22-Step by step hardening regime: heating at 500°C for 3 hours + 520°C for 3 hours + aging at 150°C for 35 hours.

3. Mechanical Properties of the AL20 Alloy

The effects of low temperatures on the mechanical properties of the AI20 alloy in annealed condition after casting are given in Table 76.

Table 76

Mechanical properties of AL20 alloy at low temperatures (according to S. Ye. Belyayev)

(1) Вид образца	Темпе- ратура испытания °C	^а р кг/мм³	S _к кг/мм³	ψ. %	8. %	obn.	S _K n
(2) Гладкий	+20 -40 -70	17,2 17,6 18,6	17,7 17,7 18,6	2,1' 1,0 0	0,5 0,2 0		
(3) С надре- зом	+20 -40 -70	16,1 16,2 16,6	16,1 16,2 16,6	0 0 0	=	0,935 0,915 0,890	0,910 0,915 0,890

(5)Примечание. $\frac{S_{\kappa_{\Pi}}}{S_{\kappa}}$: $\frac{\sigma_{b_{\Pi}}}{\sigma_{b}}$ — коэффициенты действия надреза.

Key: 1-Type of sample; 2-Smooth; 3-Notched; 4-Test temperature ${}^{\circ}$ C; 5-Note. ${}^{\circ}$ S_k ${}^{\circ}$ b_n - These are the coefficients of notch effect.

C

Tests were carried out for tension and mechanical shock resistance at a temperature of +20, -40 and -70°C on small and notched test specimens 4

The samples were cooled down to a temperature of -70° with a mixture of carbonic acid and acetone and were then held at the test temperature for 15 minutes.

Fatigue tests were carried out at a vibration frequency of 20 x 166 cycles in the case of testing at normal temperature destroy fatigue data for the AL20 alloy in comparison with other alloys is given in Table 77.

With respect to fatigue strength at 20°C the AL20 alloy is not inferior to the AL4 alloy and it surpasses the latter in ultimate strength at 250°C (see Table 74).

Table 77

Fatigue test data for AL20 alloy in comparison with other alloys (sand cast test specimens)

		/ /							
(l)	Предел устаности сплава при температуре. °C о 2,2 кг/мм ³								
88	2	0	250						
Марка сплава	(3) литой 20-10° циклов	термически обработан- ного (2·10' циклов)	терынгоски обработан- ного (2·10° циклов)						
ALI AL4 AL5 AL20	<u>-</u> 7	6,5 7,0 7,5	4,5 4,5 6,5						

Key: 1-Alloy designation; 2-Alloy's fatigue limit at temperature, $^{\circ}$ C, $\sigma_{0.2}^{\circ}$ kg/mm²; 3-Cast, 20 x 10⁶ cycles; 4-Heat treated (2 x 10⁷ cycles).

The tests were carried out under the directorship of Ye. S. Belyayev.

² The tests were carried out under the directorship of L. S. Zhukov.

in T

char

	марка	о 0 3a 10 прн 2 кг/л	50° C 1	о (2) за 100 час. при 300° С кг/мм ²			
,	сплава	(3)	термиче Е ски обра- ботанный	(9)	термиче-		
•	ALL ALA	_	6,2	_	3,0		

Creep test data for AL20 alloy in comparison with other alloys

Key: 1--Alloy designation; 2-- $\sigma_{0.2}$ for 100 hours at 250°C kg/mm²; 3--Cast; 4--Heat treated; 5-- $\sigma_{0.2}$ for 100 hours at 300°C kg/mm²; 6--Cast; 7--Heat treated.

The minimum creep values for the AI20 alloy (with $\sigma_{0.2} = 100$ hours) in comparison with other alloys are given in Table 78. Data for prolonged tension and compression tests for the AI20 alloy on individually sand cast test specimens are given in Table 79.

Table 79

Data for tension and compression tests of AL20 alloy test specimens

•	(1)	(2)	(3)	₩. 			F _K		(6) -	че	Предо	л тек; г/мм²	у-
	Состояние сплава	Вид- яспытания	ов. ке/жж³	S _K , Ke/KK	8. %	ф. %	Испытание рения в %	% •d	^а пр. кг/жж	90,2	0.1	0.05	Е. кг/жж
(8)	Отожжен- ный	(10) На растя- 11 жение	19,0	19,5	0,5	2,4		1,4	7,8	15;0	13,3	12,0	7080
	при 250° С в течение	На сжатие	116,7	5,12	_	_	56,0	-			_	_	6800
(9)	5 час После за- калки и	(12) На растя- 13жение	25,3		_	_	-	-	13,6	24,7	_	21,7	7080
	старения	На сжатие	76,8	38,7	_	_	48		_				

Key: 1--Condition of alloy; 2--Type of test; 3--T_b, kg/mm²; 4--S_K, kg/mm² ture; 5--acceleration test in % E_K; 6--Tproportionality kg/mm²; 7--Yield stress, kg/mm², at; 8--annealed at 250°C for 5 hours; 9--after hardening and aging; 10--tension; 11--compression; 12--tension; 13--compression.

The AI20 alloy is considerably superior to the AIA and AL5 alloys and is practically equal in strength to the AII alloy.

Short time testing of the AL20 alloy for portion and shear are given in Table 80.

Figure 93 shows the microstructures of the AL20 alloy. Comparative characteristics of the basic alloys are given in Figures 94 and 95.

Table 80

Results of portion and shear tests

	(2) ^m	···(3)."	C1114 - 411		''('4')'''	(5) ^{pc}	_(0)_	(7)	(8)
(1) Состояние сплава	тр = 12 кг/жж с учетом пласти-	тр = 16 кг/мм [*] с учетом упруго-го кручения	$\tau_{\mathbf{g}} = 12 \ \kappa z / \mu \mu^{\mathbf{g}}$	တ ၊	т = 12 кг/ни ⁸ максимальное, ка- сательное напря- жение	т = 16 кг/мм³, максимлльное из- пряжение	Срединй угол эа- кручивания, град	сред касательное напряжение при срезе	Модуль сдинга при кручении Е кг/ям*
(9)Отожженный при 250° С	4,1	5,5	7,9	10,4	14,5	19,4	110	14,5	2660
в течение 5 час. (10)Закаленный и состаренный	_	11,9	. —	17,7		25,6		17,6	2600

Key: 1--Condition of alloy; 2-- $\tau_p = 12 \text{ kg/mm}^2$ with consideration of plastic tortion; 3-- $\tau_p = 16 \text{ kg/mm}^2$ with consideration of elastic tortion; 4-- $\tau_p = 12 \text{ kg/mm}^2$ maximum, tangential stress; 5-- $\tau_p = 16 \text{ kg/mm}^2$ maximum stress; 6--average angle of tortion, degrees; 7-- $\tau_p = 16 \text{ kg/mm}^2$ maximum stress; 6--average angle of tortion, degrees; 7-- $\tau_p = 12 \text{ kg/mm}^2$ average tangential stress at shear; 8--shear modulus in tortion E kg/mm²; 9--annealed at 250°C for 5 hours; 10--hardened and aged.

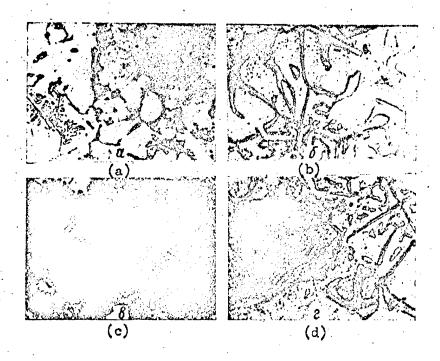
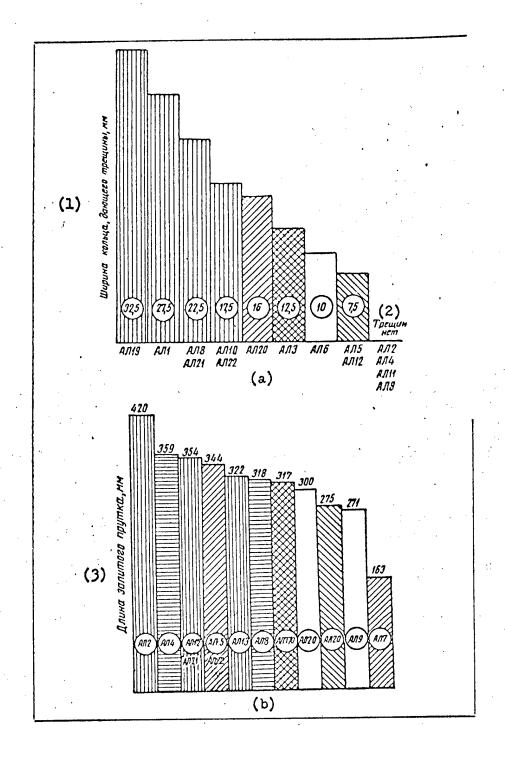


Figure 93. Microstructure of the AL20 alloy: $1 - 300^{\circ}$ C; $2 - 350^{\circ}$ C a - in cast state, 100X; b - the same, 500X; c - in hardened and annealed state, 100X; d - the same, 500X.

It was established as a result of the examination that the AL20 alloy surpasses the AL1 alloy with respect to prolonged heat resistance in the cast state but it is practically equal to the latter in the heat treated condition. The AL20 alloy is less inclined to crack formation during crystallization and quenching than the AL1, AL7, AL8, AL9, AL21 and other alloys (See Figure 94a).

The flowability of the alloy at 730-780°C is fully adequate for the casting of complex and irregular parts (Figure 94c).

The ability to retain its strength in relation to cross section thickness is higher than in the cases of the ALL, ALA, ALA, ALB, ALB and ALL2 alloys (Figure 94b).



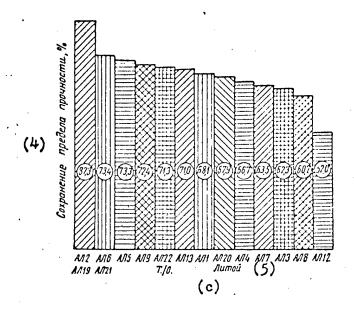


Figure 94. Comparative characteristics of the alloys: a - hot shortness of All9 and Al20 alloys in comparison with other alloys; b - (retention of tensile strength in the alloys with an increase in the casting diameter from 15 to 60 mm; c - flowability of alloys at 700°C temperature. 1-width of ring producing cracks, mm; 2--no cracks; 3--length of cast bar, mm; 4--retention of tensile strength, \$\pi\$; 5--cast.

The parts made out of the AL20 alloy are distinguished by a high air tightness (water pressure above 100 at. is withstood without failure).

The coefficients of thermal expansion and heat conduction at high temperatures are practically the same as those in the ALA and AL5 alloys.

The mechanical properties at room temperature in the cast and heat treated states are practically the same as those in many alloys (AL6, AL9 and others) with the exception of a decreased elongation in cast state.

The optimum heat treating conditions which have been developed for the AL20 alloy assure a relatively high level of heat resistance with retention of satisfactory strength at room temperature. The following heat treatment conditions are recommended for parts which operate at elevated temperatures.

Heating to below hardening at 515° C for 3-5 hours, water quenching 80-100°C and aging at 250° for 5-10° per hour. The given regime provides for a tensile strength in the alloy of 25 kg/mm² and a relative elongation of not less than 1.0% as well as a rupture strength for 100 hours at 300° of 5 kg/mm².

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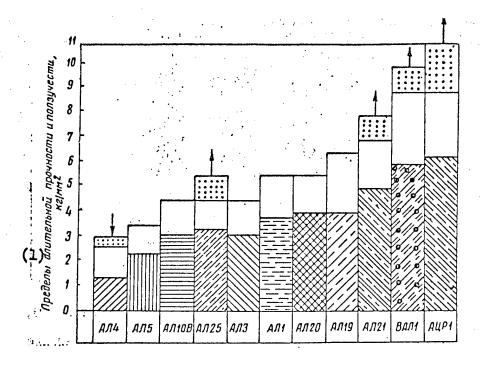


Figure 95. Rupture strength and creep limits for the alloys at 300°C. The creep limit is cross hatched; the ultimate strength is cross hatched and unshaded; increase in the rupture strengths at optimum chemical composition — dots with the arrow pointing upwards; the same, reduction — arrow pointing downwards.

1—Rupture strength and creep limit, kg/mm².

CONCLUSION

A generalization of the source material as well as the comprehensive material in the present work make it possible to draw a conclusion concerning the possibility of the development of still more heat resistant aluminum casting alloys than the alloys shown in Table 5 for prolonged services at temperatures of 0.8-0.85 from their absolute melting point. The following propositions have to be considered in the development of the new alloys:

- l. A binary system should be taken as the basis for the development of a new highly heat resistant casting alloy. This system should have: a) a eutectic with a high melting point; b) a second phase which practically does not react with aluminum at working temperature.
- 2. The melting point of the new alloy should be above 600°C which will enable to increase its working temperature to 400-500°C.
- 3. The alloy should contain not less than 35% of the eutectic and have a relatively narrow interval of crystallization (not more than 30°C) in order to assure high casting properties and to obtain extra airtight parts.

4. The basic alloying constituents should be the elements of the transition group of the periodic table in order to obtain a relatively stable desolid solution and a high temperature eutectic. The most suitable are those elements which have a low coefficient of diffusion in aluminum, a high melting point and the capability of forming complex and dispersed decomposition products at working temperatures which then create a stable microheterogeneity of the second quarter within the grains. For example an alloy intended for prolonged service at 400°C should be alloyed by the following most effective constituents: cerium, manganese, chromium, vanadium, zirconium and other analogous elements.

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- 5. The higher the alloy's working temperature to a greater degree is it necessary to strengthen the grain boundaries with phases which are stable at working temperatures [for instance AlaCuCe, AlaMnaCe, Ala(NiCu)2, AlaCuaNi] which crystallize in a branched form and for all practical purposes do not react with the a solid solution. Generally these phases form a stable lattice blocking the grains of the solid solution.
- 6. The sizes of the particles of the second phases as well as their quantity should not be very high so that a concentration of stresses, leading to a reduction in the alloy's ductility, would not occur.

The size and quantity of the particles of the second phase, optimal for a given alloy and service conditions, are found experimentally.

Data are given below which characterize the heat resistance of the alloy composition as a function of the nature of the alloying constituents. Silumin type alloys can operate for a long time only at temperatures up 200-275°C on account of the increased diffusion mobility of the silicon and aluminum alloys. Silicon does not form complex compounds with aluminum therefore the formation of the silicon particles proceeds rather rapidly during the decomposition of the & solid solution. When the Silumin type alloys are alloyed with elements of the transition group their heat resistance is raised considerably. This is explained by the strengthening of the & solid solution and the grain blocking force owing to the formation of a stable framework of complex phases. Examples of similar alloys are the alloys AL25 (ZhLS-1) and AL26 (VKZhLS) which are used for the pistons of automobile and tractor engines. It is of advantage to employe the systems Al-Mg and Al-Cu as the base when developing a high strength alloy for brief service at elevated temperatures or one which is subjected to the prolonged action of low gas or fluid pressures at temperatures which are below the aging temperatures. The alloys should have a fine grain structure with a minimum quantity of particles of second phases distributed among the grain boundaries. The degree of supersaturation of the & solid solution of the alloy by the alloying constituents should be maximum.

The most suitable systems are the Al-Ce, Al-C and Al-Ni as the basis for an alloy intended for prolonged service at temperatures around 400°C. The alloy structure should be multiphase and particles of the second phases should reliably block the grain boundaries of the solid solution. A solid solution should answer the requirements set forth in section 4.

The role of the transition type alloying constituents, increasing the interatomic bond and stability of the δ solid solution of aluminum, can be evaluated by Raynor's data (See page 64). The pausitive effects of the transition elements on the heat resistance of the alloys has also been noted by M. V. Zakharov (page 64).

The data of our investigations confirms the very positive effect of the transition group elements on the heat resistance of aluminum alloys (page 64).

Their effects can basically be reduced to the following: a) strengthening of the interatomic bond owing to the complexing of the electron structure of the multicomponent solid solution (a partial filling up of the electron shells by valance electrons, for example d in the case of atoms of manganese, chromium, nickel and others) as well as the low mobility of the transition element atoms. Presence of metallic atoms with a low coefficient of diffusion in the solid solution promotes the appearance of Cottrell "atmospheres," inhibiting the movement of dislocations; b) the formation of dispersed particles of complex phases, having a favorable effect upon the origination of microheterogeneity within the grains of the solid solution and stable at working temperatures; c) strengthening of the grain boundaries by second phases which crystallize in a branched form. These considerations found confirmation in our examinations the result of which was the development of the ATSR2 alloy operating for a long time at temperatures up to 450°C.

APPENDIX 1

Effects of phase composition on the heat resistance of basic aluminum casting alloys

дли- проч- 2/ж.и³ час. ппера-	300	ທ _ີ	3.	3,5
Предел дли- reльной проч- ности. кг/им³ за 100 час. при темпера- туре. °C	250	7	3,5	2—6
при 20° С.	ks/s ap	52	16—18	23.
(6) Характеристика микроструктуры сплава		Вторые фазы кри- сталлизуются в ви- де эвтектических прослоек. Фаза Т (Al ₆ Cu ₃ Ni) имеет разветвлен- ную форму, пре- иятствуя деформа- ции зерна д-твер- дого раствора	Модифицированная структура менсе жаропрочна, чем немодифи-	Сгрукура более многофазиа, чем в сплаве АЛБ
Степень взанио- действия вторых фаз с а-твердым рост- вором при темпе- ратуре закалки		фаза S полностью реагирует с тевр- дым раствором, тогда как фаза Тип лишь частич- но	частицы кремния быстро сфероиди- зируются	Частицы кремния сфероидизируются, фазы W (Al, Mg, Cu, Si,) и AlSi Anfe измения как фазы Mg, Si, CuAl, реагируют полиостью
нка тво-	состоянин	Сумма 77спрую- . щих элементов 14 %Си + 1,5% Mg в отношении 2,61:1) создает устойчивый пересыщенный — а-твердый раствор	Малоустонивый обедненный твер- дый раствор (с со- держанием до	Si, (ME2)Si Медь и магний в отношениях при содержании Сu (2,65:1) образована верхнем пределения фаза СuAl2) мивый а-твердый даствор
(3) Наиболее типичный фазовий состав сплавов (кроме а-твердого, раство-	Ž.	S (Al ₂ MgCu), T _{N1} (Al ₃ Cu ₃ Ni) (SI (Al ₄ Si ₂ Fe)**	Si, (115,)Si W (Al, Mg, Cu, Si,) (при содержании Си на верхнем пре- деле образуется фаза CuAl,) AlSiMnFe
(2) Химический состав (остальное алюми-	Anna Anna	3,75-4,5% Cu, 1,25-1,75% Mg, 1,75-2,25% Ni	10—13%	4—5% Si, 1,5—3,5 % Cu 0,35—0,6% Mg, 0,6—0,8% Mn
G sasana a	уврки	N TILV	АЛЗ	АЛЗ

solid solution whereas the TN1 phase reacts only partially; 11 .- Second phases crystalsolution in hardened state; 5-degree of reaction of second phases with & solid solution at hardening temperature; 6-Characteristic of alloy's microstructure; 7--6 at 20°C, kg/mm²; 8--rupture strength, kg/mm² for 100 hours at temperature, °C; 9--rhe sum of the alloying constituents (½ Cu + 1.5% kg in ratio 2.51 : 1) creates a stable phase composition of alloys (besides & solid solution); 4--Characteristic of & solid 1--Alloy designation; 2--Chemical composition (remainder aluminum); 3--Most typical lize in the form of eutectic interlayers. The T(AlcCuzNi) phase has a branch form and supersaturated A solid solution; 10 -The S phase completely reacts with the A Key:

Key to Appendix 1 (continued)

preventing deformation of the grain of the solid solution; l2-Slightly stable and deficient solid solution (with a content of up to 1.5% Si); l3-Silicon particles sphereodize rapidly; l4-modified structure less heat resistant than the unmodified; l5-Si, Mg2Si W($\Lambda l_x Mg_5 Cu_4 Si_4$) (with dopper content on upper limit the CuAl2 phase is formed); l6-Copper and magnesium in ratios for formation of S phase (2.65:1) create a more stable λ solid solution; l7-Silicon particles spherodize, the W($\Lambda l_x Mg_5 Cu_4 Si_4$) and AlSiMnFe phases change only slightly whereas the Mg2Si and CuAl2 phases react completely; l8-Structure more multiphase than that in AL5 alloy;

Предел длн- ельной проч- ости, кг/мм³ за 100 час, при темпера- туре, °C	300	3,0.	3,5	3.0
Предел дли- тельной проч- ности, кг/ми ³ за 100 час, при темпера- туре, °C	250	45	55	4,0
ири 20° С	ks/ aP	24	. 54	17
Характернстика микроструктуры сплава		Наличне Марган- цовистой фазы повышает жаро- прочность сплава	устойчивый са-твердый раствор и многофаз-ность обусловливаног более высокую жаропрочность сплава, чем у АЛ4	Кроме двойной эвтектики а + Si содержит еще и тройную а + Si + CuAl ₂
Степень взимо- действия вторых фаз с «-твердым раст- вором при томпе- рлууре закалки	•	Частиць Кромния сферондизируются; фаза Мg-Si реагирует полностью; фаза AlSiMnFe является всема устойчивой	Частицы кремния сфероидизируются, фаза СиА!, реа-гирует полностью, фаза W частично,	(26) Частицы кремния сфероидизируют-ся; фаза Сидъре реагирует неполностью; фаза Аl, Cus Fe практически не изменяется
Харакеристика «твердого раствора в закаленном состоянии		(19) «-твердый раствор недостаточно устойчив	(22) легирован боль- ше, чем в спла- ве АЛ4	(25) а.твердый ра- створ устойчивее, чем у сплава АЛТ
Нанболее типичный фазовый состав сплавов (кроме с-твердого раство-	•	Si, Mg,Si, AlSiMnFe	Si, CuAlz, W (Al _z ^{Mg} s,Cu _s Si ₄), AlSiFe	Si, CuAls, Al,Cu ₂ Fe
Химический состав (остальное влюми- ний)	•	8—10,5% Si 0,17— 0,3% Mg, 0,25— 0,5% Mn	4,5 —5,5% Si, 1,0 —1,5% Cu, 0,35—0,6% Mg	4,0—5,0% Si, 2—3% Cu
жа сплава	deW	AЛ4	AJIS	АЛ6

the CuAl2phase reacts incompletely; Al7Cu2Fe phase practically unchanged; 27--in addition to binary eutectic λ + Si a third phase λ + Si + CuAl2 is contained; solution is more stable than in the AL? alloy; 26 .- Silicon particles spheroidize; acts completely; W phase only partially; 24--Stable & solid solution and multiphasness promote a higher heat resistance in alloy than in the AL4; 25-- & solid manganis phase increases alloy's heat resistance; 22--4, solid solution is more alloyed than in AL4 alloy; 23--Silicon particles spheroidize; CuAl2 phase re-19--8, solid solution insufficiently stable; 20--Silicon particles spheroidize; Mg2Si phase reacts completely; AlSiMnFe phase is very stable; 21--presence of Key:

•			
Предел дли- тельной проч- ности, ке/миз ла 100 час. при темпера- туре. °C	3.0	1,5	3,55
Преде тельно ности, за 10 при те туре	4,2	2,5	4
25/mms 2P ubh 50° C	24	30—40	20
Характеристика микроструктуры сплава	Грагица зерна а-гвердого ра- створа укреплены слабо, что пони-жает жаропроч- пость сплава	(34) катвердого раствора укреплени слабо	Частицы фазы Мазы трудно обнару-живаются даже при увеличении в 500 раз
Степень взанио- действия вторых фаз с «твердым раст- вором при темпе- ратуре закалки	фаза СиА, пол- ностью переходит в ствердый ра- створ; фаза АІ,Си, Fе лишь частиню, частицы Si сфероидизиру- ются	Частибы в (A13Mg,) быстро коагули-руют. В этом случае сплав имеет пониженную пластичность	Частиць Кремния сферопдизируют- ся. Фаза Мд.Si полностью пере- ходит в α-твер- дый раствор
Характеристика «-твердого раство- ра в закаленном состоянии	(29) α-твердый раствор достаточно ле- гирован медью (до 5%) и устой- чивее, чем у сплавов АЛ4	(32) Сильно пересы- щенный а-твер- дый раствор обусловливает низкую жаро- прочность сплава	а-твердый ра- створ недоста- точно устойчив
Наиболее типичный фазовый состав сплавов (кроже с-твердого растворов	CuA12, Si, A1,Cu ₂ Fe	β (Al ₃ Mg ₂), Mg ₂ Si, Al ₃ Fe	Si, Mg.Si, Al ₄ Si,Fe
Химический состав (остальное алюми- ний)	4—5% Си, Si добавляется для литья в кокиль	9,5—11,5% Mg	6-8% SI, 0,2-0,4% Mg
Марка сплава	AJ17	AJ18	АЛЭ
		ı	•

ciently alloyed with copper (up to 5%) and more stable than in AI4 and AL5 alloys; 30--CuAl2 phase transforms completely into & solid solution; Al7Cu2Fe phase only partially; silicon particles spheroidize; 31--grain boundaries of & solid solution are weakly strengthened which lowers alloy's heat resistance; 32-strongly supersaturated & solid solution; 37--Particles of Mg2Si phase are very fine, they are difficult to observe even with a magnification of 500; casting in permanent mold; 29 -- & solid solution suffi-In this case alloy has a lowered ductility; 34--Grain boundaries coalesce rapidly. In this case alloy has a lowered ductility; 34--Grain boundaries of & solid solution are weakly strengthened; 35--& solid solution insufficiently stable; 36--Silicon particles spheroidized; Mg2Si phase transforms completely into in alloy; 33-- \$ particles (Algaga) & solid solution causes a low heat resistance 28--4-5% Cu, Si is added for

т дли- прот- кг/ни ⁸ учес. чисра-	300	3,0	6,57	_
Предел дли- тельной прот- ности. кг/жмв за 100 час. лри темпера- туре. С	250	4.4.	21	
им _в	K2/ QD	<u> </u>	36	_
Характеристика микроструктуры сплава		(40) Структура достаточно гетерогенная	(43) Микрогетеро- генность внутри зерен выражена сильнее, чем в других сплавах	
Степень взаимо- действия вторых фаз с а-твердым рост- вором при темис- ратуре закалки		(39)	(42) фаза СиА!, реагирует пол- ностью, коли- чество фазы Т в закаленном состоянии боль- ше, чем в литом	
Характеристика а-твердого раство- ра в закаленном	СОСТОЯНИЯ	(38) α-твердый ра- створ более устойчив, чем в сплаве - AЛ8	(41) а-твердый ра- створ значитель- но устойчивее, чем в сплаве АЛ7	
Наиболее типичный фазовий состав сплавов (кроме в-твердою раство-		Mg,Si, β (Al ₃ Mg ₂), AlSiMnFe	CuAl _s , T (Al ₁₂ Mn ₂ Cu)	
Химический состав (остальное алюми-		4,5—5,5% Mg, 0,8—1,3% Si, 0,1—0,4% Mn	4,5-5,3% Cu, 0,6-1,0% Mn, 0,2-0,4% Ti	
\$86RG2 \$3	ндвМ	АЛІЗ	АЛ19	

heterogeneous; 41--> solid solution is much more stable than in AL7 alloy; 42--CuAl2 phase reacts completely, quantity of I phase in hardened state is greater than in east state; 43--ificroheterogeneity within the grains is expressed more intensively than in 38-- Asolid solution more stable than in ALS alloy; 39-- \$ phase (AL3 Mg2) reacts completely and Mg2Si and AlSiMnFe phases are very stable; 40--Structure sufficiently other alloys; Key:

		•	
Предел дли- ельной проч- ости, кг/им ^в , за 100 час. при темпера- туре, °C	300	9	8
Предел дли- тельной проч- ности, к.г/иля за 100 час, при темпера- туре, °C	250	∞	10—12
им _в при 20° С	ks/ ap	21	22
Характеристика микроструктуры сплава		(47) Зерна «-твер- дого раствора Слокированы устойчивыми же- лезосодержащими и другими фа- зами	Зерна а-твердо- го раствора почти полностью бло- кированы устойчивыми фа- зами
Степень взапмо- действия вторых фаз с ствердым раст- вором при темпе- ратуре закалки		(46) фазы S (Al ₂ CuMg) и Mg ₂ Si реаги- руют полностью, тогда как другие фазы устойчивы	фаза S(M12CuMn) полностью реа- гирует, тогда как остальные фазы практически не реагируют
Характеристнка с.твердого раство- ра в закаленном состоянии		(45) α-твердый ра- створ более ус- тойчив, чем в сплавах типа силумин (АЛ2, АЛ3, АЛ4, АЛ5, АЛ9 и др.)	а-твердын раствор болсе устойчив, чем у всех остальных литетейных алиминевых сплавов
Наиболее типичный фазовый состав сплавов (кроме а-твердого раство-		(44t) S(Al ₂ CuMg); Al ₃ Ti; Mg ₂ Si n(Al ₈ Si ₆ Mg;Fe) или Al ₈ SiFe и фазы, со- держащие Мп и Сг	(48) S(Al ₂ CuMg), Al ₃ (CuNl), и фазы, со- держащие Мп и Сг
Химический состав (остальное алюми- ний)		3,6 —4,5% Cu, 0,7 —1,2% Mg, 1,5 —2,0% Si, 1,3 —1,5% Fe, 0,05—0,1% Ti, 0,15—0,25% Mn	4,5-6,0% Cu, 2,6-3,6% Ni, 0,1-0,25% Cr, 0,2-0,3% Mn
рке сплаве	le?V	Ал20	AJ121

фазы образуются, когда сплавы содержат примесь железа: фаза Mg_SI в сплавах типа АЛ8 образуется в том случае. прочности относится к модифицированным сплавам. При содержании Мп и Си на верхием пределе •• Более низкое значение длительной прочности относится к модифицированиым сплавам. При содс сплав АЛЗ обладает более высокими показателями. ••• Более высокое значение предела длительной прочности относится к сплаву, содержащему молибден. когда имеется при

44--S(Al2Cuig); Al3Ti; Eg2Si V (AlgSigEFe) or AlgSiFe and phases containing manganese and chromium; 45--, solid solution is more stable than in Silumin type alloys (AL2, casting alloys; 50--The S(Al2Cuin) phase reacts completely whereas the remaining phases practically do not react; 51-Grains of Assolid solution are almost completely blocked AL3, AL4, AL5, AL9 and others); 46 -- The S(Al2Cuig) and Mg2Si phases react completely stable iron bearing and other phases; 48 -- S(Al2Cuig), T(Al6Cu3Ni)2 and phases containing Win and Cr; 49-- & solid solution is more stable than in all other aluminum while the other phases are stable; 47 -- Grains of is solid solution are blocked by by stable phases. Key:

Footnotes for APPENDIX I

- * Iron bearing phases are formed when the alloys contain an iron addition; the Mg2Si phase is formed in the case when there is an addition of silicon.
- ** The lower value for the ultimate strength pertains to the modified alloys. With a maganese and copper content at the upper limit the AL3 alloy has higher indexes.
- *** The higher values for the rupture strength pertain to an alloy containing molybdenum.

Typical mechanical properties of aluminum casting alloys as a function of heat treatment conditions at room and elevated temperatures

(1) Maps	(2) Режим	Механ	Механические сво	свойства при 20°	20°C	о _в при кр	q_b при кратковременном разрыве, ка/и M^4 при температуре, °C	м разры-	1000.	9 _{100°} (2) мм. при темпе- ратуре. °C	н темпе-	Остаточная де- формация 0.2% за 100 (час.)
сплава	кой обработки	ap KZ/KK®	90.2 K2/MW	8.%	HB KZ/MM³	200	250	300	200	250	300	при температуре 300°C 302/100° кг/жи ³
ALT	T5 T1	25,0 22,0	20,0	0,6	001 06	18,0	16,0	14,0	13,0	7,0	5,5	3,7
ALZ	T2	. 16,0	0,6	- 5,0	20	15,0	13,0	8,0	7.0	4,0	2,8	1,2
ĀL3	. T1 . T2 . T5 . T7	20,0 18,0 24,0 21,0	17,0	1,0	70 65 75 70	18,0	15,0		, 1 1 6 1	11%1	3,75	2,5
AL/4	T1 T5 T6	18,0 22,0 24,0	14,0 17,0 18,0	2,0 4,0 3,6	65 70 75	16,0	14,0	10,0	8,0	5,0	2,8	
ALS.	T1 T6 T7	18,0 24,0 20,0	15,0	0,8	82.55	18,0	15,0	10.0	0.6	ا بي	3,5	2,4
AL6	12	17,0	11.0	2,0	. 55		.	-				1
£17	T4 T5	24,0 26,0	16,0	3,0	88	18,0	14,0	10,0	0,01	6,0	3,0	11

1--Alloy designation; 2--Heat treatment conditions; 3--Hechanical properties at 20°C; 4-- 0 at temporary rupture, kg/mm² at temperature, °C; 5--1000 kg/mm², at temperature, °C; 6--Permanent set 0.2% for 100 hours at temperature of 300°C, 0 3990nno ture, °C; 6-Permanent kg/mm²; 7--BiBI, kg/mm²; Key:

APPENDIX 2 (continued)

		5 1	. ,	i		ļ	1	1		}	1	1		١
Остаточная де- формация 0.2% за 100 (час.)	при температуре	300° C 30.2/100°	1,0	1,2	1		1	ı	1	4.0		4.0	. 1	5,0
	1	300	1,5	2,8	1			1	5.5	6,5	2,0	5,5	1	7.0-8.0
9100° KZ/MM³, Прн ТСМПС-		250	4,0	4,5	l		1	1	19.0	12,0	5,0	0,6		12,0
°100° K2	` -	200	8,0	6,0	1		1	ı	16.0	16,0	0,6	14,0		18.0
pasper-		300	0,6	0.6		- -	10.0	1		14.0	14,0	13,0		16,0
ременном и температ	ייייייייייייייייייייייייייייייייייייייי	250	15,0	0,11		- -	13,0	!	- -	0,61	18,0	16,0		20,0
4b при кратковременном разры-	KZ/KK up	200	22,0	14,0	-	- -	15,0	_ <u></u> .	- -	26,0 26,0	20,0	20,0		21.0
0	+	HB K2/HW ³	06	50	-\	3 -	75	88	- -	 001	8	80.		885 25
ства пон 2		% %	10,0	3.0		2,0	1,5	3,0		0.4	4.0	8.0		0.7
Secure and the 10 20°	deckine clos	°0,2 K2/KK3	17,0	14.0	2	15,0	13,0	11,0		18.0 25.0	18.0	23,0		25.0
	Механк	8 4 7 C	30,0	19,0	0,44	22,0	18,0	17.0	-	32,0 36,0	26,0	30,0	2:1	21.0 30.0 22.0
	Режим	термичес- кой обработки	T4	17.	- - c	T2	T2	13	:	T4 T5	T4	T6	=	T2 T6 T7
-		марка сплава об	АЛ8	АЛЭ	_	АЛП	A.7112	- LI 4	2174	AJ119	Ал22	A.7120		АЛ21

				-								
Марка	Режим Марка термичес-		Механические свойства при 20° C	ойства при	20°.C	о _в при кј ве. кг/жж	затковремени в при темпе	«в при кратковременном разры- ве. кг/жм² при температуре. °С		⁴ 100° кг/мм ⁸ при темпе- ратуре °C	темпе-	Остаточная деформация 0.2% за 100 (час.)
сплава	кой обработки	°b кг/жж²	°0.2 кг/жж³	5. %	HB KE/MM ⁸	200	250	300	200	250	300	при температуре 300 °C 902/100° кг/жж
8)в.л.1	TS	28,0	28,0 22,0	2,0	2,0 100	25,0	20,0	15,0 18,0 15,0 10,0	18,0	15,0	10,0	6,0
9)ацр-1	TI	22,0	16,0	1,5	. 75	0,81 0,61	18,0	17,0	17,0 17,0 15,0	15,0	11,0	6,0
					•	•		-	-	-		_

(10)Примечания:

8--VAL-1; 9--ATsP-1; 10--Notes: 1) The presented data concerning the mechanical proper-When the magnesium content is at the lower limit maximum ductility at minimum strength is assured in the above indicated duration of holding. 5) The testings at elevated temperatures were carried out by the alloys. A change in strength can attain 30% and a change in ductility can attain 100% in comparison with the data for a nominal composition. 2) In the modified state all alloys of the Silumin type (ALZ, ALM and AL9) have a rupture strength of 10.20% lower than in the unmodified. 3) If the AL21 alloy contains molybdenum it will then have a rupture strength of not less than 8 kg/mm² and a temperature of 300°C for 100 hours. 4) When it is necessary to obtain the strength characteristics of the alloys which are generally accepted methods, viz: a) short time tensile tests were carried out on test An increase or decrease in the content of the basic alloying others) or various artificial aging regimes or by varying the heating temperature and under fixed temperatures for 100 hours without failure; c) the creep limit was determined on individually east test specimens at a 300°C temperature with a 0.2% plastic should be changed by the application of various media (water, oil, salt solution and The data for the creep limit pertain to permanent set. b) the ultimate strength was determined on 10 mm diameter test specimens by tension the temperature for heating to below hardening is the same) in saltpeter heated to ties of aluminum casting alloys to attain to a nominal chemical composition of the different from those indicated in Appendix 2 the cooling rate during the hardening constituents participating in the phase transformations during heat treatment will cause a sharp change in the mechanical properties. For instance with a magnesium content at the upper limit in the AL3, ALH, AL5, and AL9 alloys it is possible to specimens of 12 mm diameter after heating for 30 minutes at the test temperature; 5) Isothermal hardening is recommended for increasing the AL21 alloy's plasticity obtain the maximum values for the tensile strengths, the yield point and h limit at a minimum value for the relative elongation. 350°C with holding of the castings for 5 hours. permanent set for 100 hours. above indicated alloys. Key:

Engineering properties and areas of application of heat resistant aluminum casting alloys

(13) Рекомендуемые области длительного применения сплавов	Порши (56) деловки цилимформи троугие детали, от которых требует- ся повышенняя герметит- ность и достаточная проч-	H H		To we buckenship metann (62) Bucokenship metann (62) Ana geranen, pagoranomik no (63)	торых требуется повышен- ная герменичность Порыши и детали, работаю- (64) пие при температурах до	пис при температурах до (05) 350° С Для терметичных деталей. рм. (66) 601ающих при температу. (66)	рах до 100 С
(12) Коррознонная стойкость и за- щита деталей от коррозни	Hohhweinas Bulue, yew y chassa All	Tobaldichtan Bulle, 4cm y change AJI 5	Повытенная	Пониженцая	•	•	Tb 2-2.5%.
(11) (клонность к казона-	Средния	(41) Bacokan (42)	Bacokan	Понижен- Средияя		CpeAux	A MOMET Gur
(10)	хорошая удовлетво- рительная	(31) Поилженная (32)	Z = 5	Huskar 35 Xopowar 36 VAODATETBOS		(39)	ямеет рисообразных крупных деталей линейная усадка может быть 2-2.5%.
(9)	удовлетво рительная (222b)	(23) Xopouran (24)	Удовлетво- рительная То же 25)		хорошая Хорошая	• •	пих деталей
	142 Понижен- ная Средняя	(15) xopoinda (16)	Средняя	, 18 Понижен- ная	(19) Tobushen-	Bucokan	разпых кру
рина кольца) жж	27.5	Не нмсет	12	32.2	22	30 He	трубооб
Жидкотекучесть Склонность и об	340.	359	344	•	320	330	литье
Линейная усад. ж. ж. %	1	0:	1,1	1.1	1.1	540 750 1.25	To Ma
"O 3 RATHR	535 740 1.35 530 750 1.15	730	730		730 1	0 75(я н в
7 7 г. винопавии у раз.	535	570			525	54	ж е ч
(1) Cnabk	ALJ AL3	AL4	AL5 AL10V	zhisi Yezbis	AL20	VALL	ATSRI 603 130 67) Примечание.

Key to APPENDIX 5

1--Alloys; 2--Temperature °C; 3--Helting; 4--Casting; 5--Shrinkage, \$; 6--Flowability at 700°C (bar test) mm; 7--hot crack formation tendency (band width) mm,

10 -- Machineability; 11 -- Gas saturation tendency; 12 -- Corrosion resistence and protection of part against corrosion; 13--Recommended areas of prolonged use of alloys; 14a--Reduced; 14b--average; 15--Good; 16--Average; 17--High; 18--Reduced; 19--Average; 20--Increased; 21--High; 22a--Satisfactory; 22b--The same; 23--Good; 24--Satisfactory; 25--Satisfactory; 26--Good; 27--Satisfactory; 28--Good; 29--Good; 30--Satisfactory; 31--Reduced; 32--Satisfactory; 33--Raised; 34--Reduced; 35-Low; 36-Good; 37-Satisfactory; 38-Good; 39-Satisfactory; 40--Average; 41--High; 42--Average; 43--High; 44--Reduced; 45--Average; 46--Very low; 47--Average; 48--Reduced; 49--Higher than in All alloy; 50--Increased; 51--Higher than in ALL alloy; 52--Reduced; 53--Raised; 54--Reduced; 55--Pistons; 56--Cylinder heads and other components from which an augmented airtightness and sufficient strength up to 275°C is required; 57-The same for the operation of the parts up to a temperature of 225°C; 58-The same for the operation of the parts to a temperature of 250°C; 59--Pistons for engines with a horsepower below 100; 60—Pistons and cylinder heads operating up to temperatures of 275°C; 61--The same; 62--Heavily loaded parts; 63--For parts operating up to 275°C requiring an augmented airtightness; 64-Pistons and parts operating at temperatures up to 325°C; 65--Pistons and parts operating at temperatures up to 350°C; 66--For airtight components operating at temperatures up to 400°C; 67--Note. The shrinkage can be 2-2.5% during the casting of tubular shaped big parts.

Physical properties	of	aluminum	casting	alloys	
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Phys:	rcar pro	opercies of atmicen		(5)
(1)	(2) Плотность г/см ²	Коэффициент термического расширения, а	Теплопроводность к кал/см-сек, °С	Электросопротив- ление, р, см.мм./м
ALL	2,75	22,3·10 ⁻⁶ (20—100° C) 24·4·10 ⁻⁶ (20—300° C)	0,31 (25°C) 0,37 (300°C)	0,0528 (20°C)
AL3	2,7	22·10 ⁻⁶ (20—100° C) 24·10 ⁻⁶ (20—300° C)	0.39 (25°C) 0.38 (300°C)	0,0449 (20°C)
AL5	2,68	23,1·10 ⁻⁶ (20—100° C) 24·10 ⁻⁶ (20—300° C)	0,38 (25°C) 0,42 (300°C)	0.0462 (20°C)
AL9	2,78	19,5·10 ⁻⁶ (20—100° C) 25,6·10 ⁻⁶ (20—300° C)	0,25 (25°C) 0,34 (300°C)	0,0595 (20°C)
VT50	2,74	18,1·10 ⁻⁶ (20-100° C) 23,6·10 ⁻⁶ (20-300° C)	0,31 (30° C) 0,35 (300° C)	0,0518 (20°C)
AL21	2,83	22,9·10 ⁻⁶ (20—100° C) 27,8·10 ⁻⁶ (20—300° C)	0,27 (30°C) 0,3 (300°C)	0;0572 (20°C)
VALL	2,89	23.8·10 ⁻⁶ (20—100° C) 28,7·10 ⁻⁶ (20—300° C)	0,32 (100° C 0,37 (300° C	0,0545 (20°C)
ATSRL	2,8	23,6·10 ⁻⁶ (20—100° C) 26,7·10 ⁻⁶ (20—300° C)	0,23 (25° C) 0,27 (300° C	0,053 (20°C)
ALLOV	2,78	22,3·10 ⁻⁶ (20—100° C) 24,4·10 ⁻⁶ (20—300° C)	0,40 (25°C) 0,42 (300°C)	0,046 (25°C)
AI21	2,72	19·10 ⁻⁶ (20—100° C 20,5·10 ⁻⁶ (20—300° C	0.38 (25° C 0,38 (300° C	0,050 (25°C)
A126	2,68	17,0·10 ⁻⁶ (20—100° C	0,40 (25° C 0,42 (300°	0,058 (025°C)
	1	1	MEN PUTERBAN	гемператур, в которо

(6) Примечание. В скобках дана температура или интервал температур, в котором определено это свойство.

Key: 1-Alloys; 2-Density g/cm²; 3-Coefficient of thermal expansion; λ; 4-Thermal conductivity λ cal/cm·sec, °C; 5-Electrical resistance, β, cm·mm²/m; 6-Note. The parentheses contain temperature or the temperature interval in which this property was determined.

Hardness of binary aluminum alloys at temperature of 300°C as a function of their composition, state (load of 100 kg, ball diameter 10 mm) and heat treatment condition

	3) ((4) T	Твердость. НВ. кг/им при содержинии втэрэгэ компонента.	. HB.	KE/KK	DDH CO	держ зн	HI BTO	X 6160	ОМПОН	ента. %		
Выдер	6 0		6	e.	+	- N	9	-	8	6	10		12
30 ce 60 m	сек. —	8,5	11	13,6		27,1	11	30,0	31,2		32,1 22,3	11	
после стабилизации 30 се час.)	сек. —	3,2	6,99	8,5	9,66	13,0	12,1 7,9	11	13,5 8,65	11	15,9	11	1.1
естественно 30 се дней) 60 м	сек. —	3,77	8,63	10,0	9,0	24,8	24,1 14,4	11	23,3 14,0	11	22,0 14,1		11
30 ce	сек. —	5,1	11	20,5	25,5 16,2	33,2	32,0 19,5	31,8	31,5	11	31,3	ij	11
30 ce .60 MI	Cek	6,62 3,56	8,44 6,62	11	9,44	13,2	13,6 8,68	11	14,4 8,87	11	15,7 8,97	11	! !
30 сек. 60 мин.	сек	11	11	24,2	11	37,7	-11	39,0 16,0		11	42,6 14,2		41.0
30 ce	сек. —	11	11	11	15,9	!!	7,9	! !	23,8 8,0	11	2,4	11	7,79
30 сек. 60 мин.	сек. —	11.	11	11	17,4 7,39	11	$\begin{vmatrix} 22,9\\8,53 \end{vmatrix}$	1 1	28,0	11	30	11	25,7 8,21
30 сек. 60 мин.	сек. —	11.	11	24,6	11	32,8 10,7	11		11	11	10,8	11	36,1

(300-100 hours); 1--Alloy system; 2--State of alloys; 3--Holding time; 4--Brinnell, kg/mm2 at content of second component, %; 5.-Cast; 6.-Cast after stabilization (300-100 hor 7-Hardened and naturally aged (10 days); 8.-Hardened and annealed (300°C - 5 hours); 9.-Hardened and stabilized (300°C, 100 hours); 10.-Cast; 11.-Cast and hours); 9--Hardened and stabilized (300°C, 100 hours); 10--Cast; 11--Cast stabilized (300°C, 100 hours); 12--Hardened and naturally aged (10 days); 13--Hardened and annealed (300°C, 5 hours) Key:

															1
Система		-					-	вердос	Твердость, НВ, кг/км	. K2/K	• ¥				
cnassos	Состояние сплавов	Быдержка	0.5	1	2	3	+	2	9	7	8	6	10	11	_12
A1—Mg	Закаленнай и стабилизи- рованный (300° С, 100 час.)	30 сек. 60 мин.	11	11	11	11	15,7 7,75	11	20,0 8,53	11	23.0 8,53	11	2,4 9,5	11	26.0 8,63
Al—Zn	Литой (15)	30 сек. · 60 мин.	11	11	11	9,23	11	11	9,27	11	11	9,33	11	11	9,77
	Литой к стабилизирован- ный (300° с. 100 час.)	30 сек. 60 мин.	11	11	11	4,7	5,05 2,0	11	11	11	5,28 2,37	11	11	11	2,0
	Закаленный и естественно состаренный (160 дней)	30 сек. 60 мин.		11	11	2,2	15,5 2,96	11	11	11	15,27 3,32	11	11	. 1 1	16,37 3,43
	3акаленный и отпущенный (300° С, 5 час.)	30 сек. 60 мин.	11	11	11	9,95	11	11	8,0 4,88	11	11	10,6	11	1 1	11,9
,	Закаленд (до) и стабилизи- рованный (300° С 100 час.)	30 сек. 60 мин.	11	11	11	11	5,0	11	11	11	5,2	11	11	11	5,5
Al—Si	Литой и стабилизирован- ный (300° С, 100 час.)	30 сек. 60 мин.	11	7,36	9,07	11	10,0	11	10,2 8,4	11	9,8	1.1	10,9 9,55	11	11,4
	Закаленћыг и естественно состаренный (10 дней)	30 сек. 60 мин.	1.	9,39 6,24	9,39 6,92	11	10,1	. 1	10,3	1	10,6	!	11,0		12.5
	Закален Б. д. нестественно состаренный (10 дней)	30 сек. 60 мин.	11	9,39	9,39 6,92	11	10,1	1.1	11	11.	11,2		10,4	11	11
	Закаленћый и отпущенный (300° С, 5 час.)	30 сек. 60 мин.	11	8,2 5,1	9,5	11	10,0	1 1	8,01	1 1	10	1.1	10,1	11	9.8

14.—Hardened and stabilized (300°C, 100 hours); 15.—Cast; 15.—Cast and stabilized (300°C, 100 hours); 17.—Hardened and naturally aged (160 days); 18.—Hardened and annealed (300°C, 5 hours); 19.—Hardened and stabilized (300°C, 100 hours); 21.—Hardened and naturally aged (10 days); 22.—Hardened and annealed (300°C, 5 hours); 22.—Hardened and annealed (300°C, 5 hours); Key:

Chrys							1	вердос	Tb, H1	Твердость, НВ, ка/им	8,3				
сплавов	Состояние сплавов	Выдержка	0,5	-	7	8	*	2	9	7	∞	6	01	=	12
	(24) Закаленный и стабилизи- рованный (300° С, 100 час.)	30 сек. 60 мин.	6,3 4,8	7,0	9,61 6,58	11	10 7	11	7,4	11	10 8,7	11	10,2 9,6	11	9,5
Al—Ni	Литой (25)	30 сек. 60 мин.	11	12,0 8,1	13,2 14,4 8,5 9,17	14,4	13,9	11	11	17,5	11	11	11	l I	11
•	(26) Литой и стабилизирован- ный (300° С, 100 час.)	30 сек. 60 мин.	11	9,08 10,48	9,08 10,48	11	11	-	11	11	11	11	11	11	11
	(22) Закаленный и естествен- но состаренный (10 дней)	30 сек. 60 мин.	9,0 5,83	10	10,4	11	11	·	- 11	11	1,1	11	11	11	
	(28) Закаленный и состарен- ный (300° С, 100 час.)	30 сек. 60 мин.		9,71 6,83	8,80 6,28	11	11.	11	11	11	11	11	11	11	111
A1—Mn	Литой (29)	30 сек. 60 мин.	11	13,3	20,2	21,5	11	11	11	11		- 1 1	1, 1	11	11
	Литой и стабилизирован- ный (300° С, 100 час.)	30 сек. 60 мин.	13,1 9,7	14,5	16	11	11	. 1 1	11	11	11	11	11	11	i i

24-Hardened and stabilized (300°C, 100 howrs); 25--Cast; 26--Cast and stabilized (300°C, 100 hours); 27--Hardened and naturally aged (10 days); 28--Hardened and aged (300°C, 100 hours); 29--Cast; 30--Cast and stabilized (300°C, 100 hours) Key:

Система		. 1						вердос	Твердость, НВ, кг/им	3, K2/M.					
сплавов	Состояние сплавов	Выдержка	0.5	_	5	e	4		<u>۔</u> و	1		6	10	=	12
Al—Mn	(31) Закаленный и естественно состаренный (10 дней)	30 сек. 60 мин.	13.9 9.04	13,9 12,53 9,04 10,7	15,9 11,56	1 1	11	11	11	11	11	11	11	1 !	11
	(32) Закаленный и отпущенный (300° С, 5 час.)	30 сек. 60 мин.	1 1	13,5	16,2	18,4	11:	.11	11	11	11	11	11	11	11
Al—Fe	. (33) Закаленный и стабилизи- рованный (300° С, 100 час.)	30 сек.	12,8 9,98	13,4	13 15,55 10,4 12,25	11	11	11	11	11	11	1 !	: 11		11
	(34) Литой и стабилизирован- ный (300° С, 100 час.)	30 сек. 60 мин.	9,98	10,24 8,5	10,24 10,9 8,5 8,05	11		:11	11	11.	11	11	. 11	11	11
	(35) Закаленный и естественно состаренный (10 дней)	30 сек. 60 мин.	9,11	10,5 11,9 8,40 8,33	11,9 8,33	11		11:	1 I·	11	11 .	11	11	.	11
	(36) Закаленный и состаренный (300° 5 100 час.)	30 сек. 60 мин.	9,41 6,11	10,24 8,55	10,24 11,9 8,55 8,65	1 1	11	1.1	11	11	11	11	- 11	11	11

31--Hardened and naturally aged (10 days); 32--Hardened and annealed (300°C, 5 hours); 33--Hardened and stabilized (300°C, 100 hours); 34--Cast and stabilized (300°C, 100 hours); 35--Hardened and naturally aged (10 days); 36--Hardened and aged (300°C, 100 hours). Key:

CSO: 9799

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